



OWNER'S & OPERATOR'S GUIDE: MD-11 FAMILY

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MD-11 family specifications

The MD-11 family is sub-divided into passenger & freighter aircraft and those with PW & GE engines.

The MD-11 was launched in 1986 as a replacement for the DC-10-30, and to provide airlines with higher capacity for growth on long-haul routes. The aircraft was also intended to have a range of about 7,000nm with a standard tri-class configuration of passengers. This would allow airlines to operate non-stop trans-Pacific routes, since it was anticipated that this market, and others, would experience liberalisation.

The MD-11's development was cut short by the initial performance shortfalls it suffered on entry into service, and from which it never really recovered. McDonnell Douglas (MDC) was able to provide performance improvement programmes to recover the originally intended performance, as well as later to increase the aircraft's performance.

The MD-11, however, was outperformed by the A340-300 which was launched in 1987, and then by the 777 launched in 1990. These two reduced orders for the MD-11 to a trickle, and caused outstanding commitments to be cancelled. Plans to develop stretch and longer-range versions of the MD-11 were scrapped, and the last aircraft were built in 2000, with total firm orders reaching 200.

The first MD-11s, belonging to American Airlines, were converted to freighters as early as 1996, with some of the aircraft being as young as five years old. American had clearly expressed its dissatisfaction with the MD-11, and quickly sold its 19 aircraft to FedEx. Korean Air also sold three of its fleet for freighter conversion at the same time. Although by this time the MD-11 had gained universal disapproval as a passenger aircraft, it became respected as a freighter, and demand from freight carriers has remained strong. Out of 200 aircraft ordered, 58 were specified as freighters and five as Combis, indicating that interest in the MD-11 as a passenger aircraft was always weak. The majority of passenger-configured aircraft have been modified to freighter, and there now remain only 26 passenger-configured aircraft potentially available for conversion to freighter (*see MD-11 fleet analysis, page 14*).

Initial development

Perhaps the MD-11's failing was the fact that it was a derivative of the DC-10, and not an all-new design. The MD-11 tri-jet features an 18-foot fuselage stretch over the DC-10, which allows a standard tri-class capacity of 298. This is a combination of 16 first-, 56 business- and 226 economy-class seats. The aircraft also offered a two-man flightdeck, a larger wing and fuel capacity than the DC-10-30, and the choice of PW4000-94 and CF6-80C2D1F engines.

The original aircraft had a maximum take-off weight (MTOW) of 602,500lbs, and fuel capacity of 38,615 US Gallons (USG). It was intended to have a range of 6,840-7,000nm with its standard load of passengers. Many early aircraft actually had their MTOWs certified at 605,000lbs.

The PW4000-94 engines available for the MD-11 were rated either at 60,000lbs thrust or 62,000lbs thrust and had full authority digital engine control (FADEC) as a standard feature. These engines are designated the PW4460 and PW4462.

The CF6-80C2 variant on offer is the -80C2D1F, rated at 61,500lbs thrust. The F suffix indicates that the engine has FADEC controls, which are standard on the CF6-80C2 powering the MD-11. The -80C2 family comes with or without FADEC controls.

Both engines are flat rated at 86 degrees Fahrenheit, equal to 30 degrees centigrade, which means that take-off thrust is held constant for an outside air temperature of up to 30 degrees, but then reduced for higher outside temperatures.

The first aircraft was delivered to Finnair in November 1990, and in early 1991 it was revealed that the MD-11's performance was short of its design specification. The aircraft had higher fuel burn, empty weight and airframe drag than the intended design specification and consequently suffered a range shortfall. PW4000-powered aircraft had 6.7-8.4% higher fuel burn and a reduced range of 6,270nm, and CF6-powered aircraft had a 4.5-5.3% higher fuel burn and a range of 6,460nm.

Besides passenger accommodation, the MD-11 can accommodate 32 LD-3

containers in its underfloor compartment. These are loaded in pairs, with 18 in the forward section and 14 in the aft. These each have an internal volume of 146 cubic feet, and so provide a total of 4,672 cubic feet (*see table, page 13*). The MD-11 also has a space for bulk freight at the rear of the fuselage which provides 510 cubic feet.

Passenger-configured aircraft have a maximum zero fuel weight (MZFW) of 400,000lbs and an operating empty weight (OEW) of 283,975lbs for the MD-11, and 291,120lbs for the MD-11ER. This gives the MD-11 a structural payload of 116,025lbs, and the MD-11ER a structural payload of 108,880lbs (*see table, page 13*). The tare weight of 6,880lbs for the 32 LD-3s should be deducted from this, giving the aircraft a net structural payload of about 109,000lbs and 102,000lbs.

A full load of passengers will have a payload of about 67,000lbs, and so the aircraft will be able to accommodate an additional freight payload of 35,000lbs.

Performance improvement

While receiving negative publicity, MDC overcame the MD-11's performance shortfalls with a series of performance improvement packages. The first step was a rise in MTOW to 618,000lbs, which was later increased in 1993 to 625,500lbs. MDC also offered a drag reduction programme, which increased the aircraft's range.

In 1994 MDC introduced the MD-11ER (extended range), which provided the aircraft with a range of up to 7,210nm. This involved a further increase of MTOW up to 630,500lbs, the option of installing two auxiliary fuel tanks which took fuel capacity up to 42,584 USG, a 1,500lbs reduction in operating empty weight and a series of aerodynamic improvements.

The majority of MD-11s now have MTOWs of 618,000lbs, 625,500lbs and 630,500lbs. Most aircraft have maintained the original fuel capacity of 38,615 USG, however, and have not selected the option of the two auxiliary tanks.

The several increases in MTOW did have the effect of improving the aircraft's range, and the aircraft with an MTOW of 630,500lbs has about a 300nm longer range than aircraft with the initial MTOW of 602,500lbs (*see table, page 13*). The effect of the single and double auxiliary fuel tanks was only to increase the range of the aircraft at the maximum fuel line portion of the aircraft's payload range curve. This means that range is only added when the aircraft is carrying a payload of about 250 passengers or fewer, which explains why few aircraft have had them installed.

Combi aircraft

Alitalia was the only airline to order the Combi variant. The aircraft had a freight door at the left rear of the fuselage which is 102 inches high and 160 inches wide. The size of the freight area on the maindeck varies. It can accommodate either two 88-inch wide or two 96-inch wide by 125-inch long freight containers at the rear, and then the same type of containers in pairs. The actual number depends on the length of the freight cabin, with two, three or four pairs of containers being provided, thereby allowing four, six, eight or 10 containers which provide 2,224, 3,342, 4,460 or 5,578 cubic feet of freight volume.

These containers result in a corresponding drop in the number of tri-class seats on the aircraft, varying from 129 to 202.

Freighter configuration

Of the 200 MD-11s built, 58 were factory freighters, although more 70 passenger aircraft have since been converted. The factory freighter (MD-11F) and converted freighter (MD-11CF) have the same specification weights and capacities for freight accommodation.

As with the passenger aircraft, the MD-11F and converted freighter can accommodate 32 LD-3 belly containers.

There are several loading options for the maindeck. The two most popular maindeck containers are the 88-inch and 96-inch wide containers, both of which are 97 inches tall and 125 inches long, and are contoured to the inside profile of the aircraft. The 88-inch wide containers have an internal volume of 568 cubic feet, and the 96-inch wide containers have an internal volume of 607 cubic feet. These provide a total of 13,632 cubic feet and 14,568 cubic feet. These are loaded in pairs, with a total of 24 containers taking up the length of the fuselage, except for two at the rear.

The two containers at the rear of the fuselage are 125 inches wide, 97 inches tall and either 88 or 96 inches long. These provide 553 cubic feet or 605 cubic feet of volume each, and so add 1,106 or 1,210 cubic feet to the maindeck volume.

Total maindeck volume is therefore 14,738 cubic feet with 88-inch wide containers, and 15,778 cubic feet with the 96-inch wide containers (see *table, this page*). The aircraft thus has a total freight volume of 19,410 cubic feet when using 88-inch wide containers on the maindeck, or a total of 20,450 cubic feet when using 96-inch wide containers (see *table, this page*).

The MD-11F/CF has a high MZFW. There are two MZFW options for the MD-11: 451,300lbs and 461,300lbs (see *table, this page*). The aircraft has an

MD-11 FAMILY SPECIFICATIONS

Variant	MD-11	MD-11	MD-11	MD-11ER
MTOWlbs	602,500	618,000	625,500	630,500
MZFWlbs	400,000	400,000	400,000	400,000
OEWlbs	283,975	283,975	283,975	291,120
Gross structural payload lbs	116,025	116,025	116,025	108,880
Fuel capacity USG	38,615	38,615	38,615	38,615/ 42,584
Seats	298	298	298	298
Range nm	6,300	6,500	6,600	6,600/ 7,210
Belly freight cu ft	4,672	4,672	4,672	4,672
Engine variant	PW4460/62 CF6-80C2D1F	PW4460/62 CF6-80C2D1F	PW4460/62 CF6-80C2D1F	PW4460/62 CF6-80C2D1F
Variant	MD-11F/CF	MD-11F/CF	MD-11F/CF	MD-11ERF/CF
MTOWlbs	602,500	618,000	625,500	630,500
MZFWlbs	451,300/ 461,300	451,300/ 461,300	451,300/ 461,300	451,300/ 461,300
OEWlbs	248,567	248,567	248,567	248,567
Gross structural payload lbs	202,733/ 212,733	202,733/ 212,733	202,733/ 212,733	202,733/ 212,733
Fuel capacity USG	38,615	38,615	38,615	38,615/ 42,584
Maindeck freight Container	96 in X 125 in	96 in X 125 in	96 in X 125 in	96 in X 125 in
Number	26	26	26	26
Volume cu ft	15,778	15,778	15,778	15,778
Belly freight Container	LD-3	LD-3	LD-3	LD-3
Number	32	32	32	32
Volume cu ft	4,672	4,672	4,672	4,672
Total volume cu ft	20,450	20,450	20,450	20,450
Container tare lbs	27,420	27,420	27,420	27,420
Net structural payload lbs	175,493/ 185,493	175,493/ 185,493	175,493/ 185,493	175,493/ 185,493

OEW of 248,567lbs. This leaves a gross structural payload of 202,733lbs and 212,733lbs.

The tare weight of containers has to be deducted from this. The LD-3s have a unit tare weight of 215lbs, and so have a total tare weight of 6,880lbs. The 96-inch wide by 125-inch long maindeck containers have a unit tare weight of 790lbs and so a total tare weight of 20,540lbs. The total container tare weight is therefore 27,420lbs. This leaves the aircraft with a net structural or revenue payload of 175,493lbs and 185,493lbs, depending on MZFW (see *table, this page*). This gives the aircraft a maximum packing density of 9.1-9.6lbs per cubic foot.

These specification MZFW, OEW and payload weights are the same for both factory and converted freighters.

Stage 3 & 4 compliance

Stage 3 and 4 compliance is an issue for operators and investors to consider.

Stage 3 compliance is the cumulative (total) of the three noise measurements, measured in equivalent perceived decibels (EPNdB), being lower than those permitted for an aircraft on its MTOW, maximum landing weight and engine number configuration. The permitted cumulative noise emissions are 306.9 EPNdB for aircraft with an MTOW of 602,500lbs, 307.3 EPNdB for an aircraft with an MTOW of 618,000lbs, and 307.6 EPNdB for an aircraft with an MTOW of 630,500lbs. These three gross weight variants have actual cumulative noise emissions 11.3-14.2 EPNdB lower than those permitted, giving them a comfortable Stage 3 compliance margin.

Stage 4 compliance requires the aircraft's cumulative noise emissions to be at least 10EPNdB lower than those permitted for Stage 3 compliance. All variants of the MD-11 therefore comply with Stage 4 emissions. **AC**

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MD-11 fleet analysis

The MD-11 fleet is divided freighters and passenger aircraft available for conversion.

The MD-11 can be sub-divided into a number of categories: aircraft powered by PW4000-94 engines; aircraft powered by the CF6-80C2D1F; passenger-configured aircraft; freighter-converted aircraft; and factory-freighter aircraft. There were 200 MD-11s built between 1990 and 2000.

Although it was rejected as a passenger aircraft, the MD-11 has proved popular as a freighter. Its relatively high maximum zero fuel weight (MZFW) gives it a high payload for its size. It is also in a class of its own with respect to its freight capacity. A large number have been converted to freighter, and fewer than 40 are potentially available for modification. The MD-11 will no doubt continue to be a popular freighter, and its limited numbers are likely to keep its market values high.

Of the 200 built, 195 are still in operation or inactive, with five having been destroyed in accidents. Out of 195 aircraft, 58 are factory-freighters, and 76 have been converted to freighters since 1996. There are also another four Combi configured aircraft.

This leaves another 57 aircraft, 23 of which are in storage or inactive, and 21 have been acquired by UPS or FedEx for conversion to freighter. There are also 34 in passenger configuration, but eight of these have already been acquired by FedEx or UPS for conversion to freighter, leaving 26 passenger aircraft that have not yet been acquired for conversion.

Most aircraft, 117 units, have CF6-

80C2D1F engines and 78 have PW4000-94 powerplants (see table, this page).

Passenger fleet

There are 17 PW4000-powered and another 17 CF6-powered MD-11s operating as passenger aircraft.

Of the 17 active PW4000-powered aircraft, eight have been acquired for conversion to freighter. Three are ex-Delta aircraft have been bought by United Parcel Service (UPS) for conversion to freighter, and are currently on lease to Ethiopian Airlines and World Airways. Another five ex-Swissair/Swiss Airlines aircraft are on short-term leases to VARIG and have been acquired by UPS for conversion to freighter.

This leaves just nine passenger-configured PW4000-powered aircraft potentially available for conversion to freighter. These include on aircraft on lease to Air Namibia, two Saudi Arabian Government aircraft and six aircraft leased to World Airways.

There are 17 active aircraft equipped with CF6-80C2D1F engines, including seven operated by Finnair, four of which were originally delivered to the airline, two were acquired from City Bird and one from VARIG. Finnair will not add to its MD-11 fleet. Another 10 are operated by KLM. All 17 are potential candidates for freighter conversion in the future.

There were also five Combi-configured aircraft built for Alitalia. Four of these remain in Combi configuration.

Factory freighters

There are 58 factory-built freighters and convertible aircraft in operation, 13 of which are equipped with PW4460/62 engines. China Cargo Airlines, a subsidiary of China Eastern Airlines, operates one, FedEx has three, Martinair has six and World Airways has two.

The other 45 aircraft are powered by CF6-80C2D1F engines. FedEx has the majority with a fleet of 18, EVA Airways has nine aircraft, Lufthansa cargo has 14, and Saudia has four.

Converted freighters

There are 76 converted freighter aircraft in operation, although the number is constantly increasing as more become modified. The fleet of modified aircraft is dominated by FedEx and UPS. The fleet is split between 35 aircraft equipped with PW4460/62 engines and 41 equipped with CF6-80C2D1F engines.

The PW4000-powered fleet includes six ex-Swissair and four ex-China Airlines aircraft operated by FedEx, five ex-China Eastern aircraft operated by China Cargo Airlines, four ex-LTU/Swissair aircraft operated by Transmile Air Service, two ex-LTU/Swissair and 10 ex-Japan Airlines aircraft operated by UPS, and single units operated by VARIG LOG, Air Canada, Martinair and World Airways.

The 41 CF6-powered converted freighters comprise a single ex-Alitalia aircraft converted to freighter and 40 ex-passenger aircraft. These 40 are dominated by 18 ex-American and three ex-Garuda aircraft operating with FedEx, four ex-Delta and VARIG aircraft operating with Gemini Air Cargo, five ex-VASP and Alitalia aircraft operating with Lufthansa Cargo, and seven ex-VARIG and VASP aircraft operating with UPS. EVA Airways operates two of their own converted aircraft and Shanghai Airlines has a single ex-EVA Airways aircraft.

Stored aircraft

Twelve ex-Delta aircraft are in storage awaiting conversion to freighter. Three of these aircraft are for UPS and nine are for FedEx. Another five ex-VARIG CF6-powered aircraft, that do not have engines, are in storage. There is speculation these will be converted to freighter for operation by Aeroflot.

There are also four ex-Thai CF6-powered aircraft stored that have been bought by UPS.

FedEx has acquired a total of 57 MD-11s, and UPS has 37. Both carriers are likely to attempt to acquire more. [AC](#)

MD-11 FLEET SUMMARY

	PW4000	CF6-80C2D1F	Total
Active passenger	9	17	26
Active passenger acquired for freighter conversion	8		8
Combi		4	4
Factory freighter	13	45	58
Converted freighter	35	41	76
Stored acquired for conversion	12	9	21
Stored/inactive	1	1	2
Total	78	117	195

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MD-11 modification programmes

MD-11 modification programmes include freighter conversion, structural modifications & weight upgrades.

There are several groups of modifications and upgrades for the MD-11. These include structural upgrades, weight upgrades, passenger-to-freighter modification and avionic upgrades.

Structural modifications

Three major airworthiness directives (ADs) have been issued for the MD-11: the engine pylon upper spar replacement; the insulation blanket replacement (AD 2000-11-02); and the inspection and treatment of the horizontal stabiliser barrel nuts (see *MD-11 maintenance analysis & budget, page 19*). These have all been completed, but required a large number of man hours (MH).

Other modifications that are common in C checks are the torque shaft bearing inspections on the passenger doors, while service bulletin (SB) MD11-53-066 requires an inspection on the left and right main landing gear doors, which takes about 40 MH to complete. There are also additional wiring inspection SBs.

Weight upgrades

Finnair took its MD-11s with initial maximum take-off weights (MTOWs) of 615,615lbs. "We upgraded the MTOWs over the years, through the application of five modifications, to the maximum 630,500lbs," says Tapio Leskinen, MD-11 fleet manager at Finnair. "This was done using three paper modifications and two structural modifications, one of which was a wing reinforcement, and the other a modification to landing gear bolts. Both modifications can be carried out during a C check. While kits are available from Boeing for adding fuel tanks in the forward cargo compartment,

we never saw the economic case for adding them at the expense of revenue freight, and so have not fitted them." It is not possible to upgrade the maximum landing weight (MLW).

Cargo conversion

The MD-11 has become a popular freighter, with strong demand from FedEx and UPS leaving only 26 passenger-configured aircraft that could potentially be modified. These include aircraft operating with Finnair, KLM and World Airways (see *MD-11 fleet analysis, page 14*).

The freighter conversion modification includes the installation of a 140-inch wide by 102-inch high main cargo door, located in the forward fuselage. Modification takes 90-120 days to complete.

The conversion work also involves the removal of passenger furnishings, installation of a cargo net or solid barrier; installation of a maindeck cargo handling system, and structural enhancements.

The converted freighter (CF) has the same weight specifications and can accommodate the same type and number

of freight containers and pallets as the factory freighter (F). The F and CF have two maximum zero fuel weight (MZFW) options of 451,300lbs and 461,300lbs. These are high relative to the aircraft's size and give it a high structural payload.

There are two main providers of MD-11 conversion, which are also sub-contractors for the Boeing modification programme: Singapore Technologies Aerospace and Alenia Aeronavali. Typically the list price for both programmes is \$12 million per aircraft. Modification to freighter will incur additional costs in respect of the freight door power system, cargo crash barrier and installation of a maindeck freight handling system.

Converted freighter

A converted MD-11 freighter has 26 maindeck positions (two are 84-inch pallets, as opposed to the 96-inch pallets, at the end of the maindeck), and 32 LD-3s in the lower hold. The aircraft has an MTOW of 630,500 lbs and an MLW of between 471,500lbs and 491,500lbs, depending upon the options chosen in the conversion programme. The aircraft has



The MD-11's attractiveness as a freighter comes from its high MZFW that gives it a high structural payload. The aircraft also has a durable airframe and will continue to operate for another 10-15 years.

PAYLOAD CHARACTERISTICS MD-11F/CF

Aircraft type	MD-11F/CF	MD-11F/CF
MZFW-lbs	451,300/ 461,300	451,300/ 461,300
OEW-lbs	248,567	248,567
Gross structural payload-lbs	202,733/ 212,733	202,733/ 212,733
Type maindeck containers	88 in X 125 in	96 in X 125 in
Number maindeck containers	26	26
Unit volume maindeck containers-cu ft	568	607
Unit tare weight maindeck containers-lbs	780	790
Total volume maindeck containers-cu ft	14,738	15,778
Total tare weight maindeck containers-lbs	20,280	20,540
Type lowerdeck containers	LD-3	LD-3
Number lowerdeck containers	32	32
Unit volume lowerdeck containers-cu ft	146	146
Unit tare weight lowerdeck containers-lbs	215	215
Total volume lowerdeck containers-cu ft	4,672	4,672
Total tare weight lowerdeck containers-lb	6,880	6,880
Total volume all containers-cu ft	19,410	20,450
Total tare weight all containers-lbs	27,160	27,420
Net structural payload-lbs	175,573/ 185,573	175,313/ 185,573
Maximum packing density-lbs/cu ft	9.04/ 9.56	8.57/ 9.07

an operating empty weight (OEW) of 248,567lbs, providing the aircraft with a gross structural payload of 202,733lbs or 212,733lbs, depending on the MZFW option selected (see table, this page).

The internal volumes of the maindeck containers vary between 524 and 605 cubic feet. There are five maindeck pallet and container configurations. These include the use of 88-inch by 125-inch and 96-inch by 125-inch container configurations, which are the two most popular maindeck containers. These are both 97 inches tall and 125 inches long, and are contoured to the inside profile of the aircraft.

The 88-inch wide containers have an internal volume of 568 cubic feet, and the 96-inch wide containers have an internal volume of 607 cubic feet. These provide a total of 13,632 cubic feet and 14,568 cubic feet (see table, this page). These are loaded in pairs, with a total of 24 containers taking up the length of the fuselage, except for two at the rear. The two containers at the rear of the fuselage are 125 inches wide, 97 inches tall and either 88 or 96 inches long. These provide 553 cubic feet or 605 cubic feet of volume each, and so add 1,106 or 1,210 cubic feet to the maindeck volume.

The total maindeck volume is therefore 14,738 cubic feet with the 88-inch wide containers, and 15,778 cubic feet with the 96-inch wide containers (see table, this page). The aircraft therefore

has a total freight volume of 19,410 cubic feet when using 88-inch wide containers on the maindeck, or a total of 20,706 cubic feet when using 96-inch wide containers (see table, this page).

The 32 LD-3 containers have a combined tare weight of 6,880lbs and internal volume of 4,672 cubic feet.

The tare weight of the aircraft's containers is 27,160lbs or 27,420lbs, depending on the container type used. This has to be deducted to result in a net structural payload. The container options provide the aircraft with a net structural payload varying between 175,313lbs and 185,573lbs (see table, this page), depending on the MZFW option and containers used.

Avionics upgrades

There are several modifications that are mandatory on all aircraft in Europe. Two sets of VHF communication transceivers must be installed with 8.33kHz frequency spacing above FL245. Additional proposed new communications rules are being considered covering 8.33 kHz, extending it to cover above FL195. Two sets of VHF communication transceivers with 25kHz frequency spacing are mandated below FL245.

Traffic collision avoidance systems (TCASs) have already been mandated.

In addition terrain awareness and

warning systems (TAWS), which are currently known as enhanced ground proximity warning systems (EGPWS), are mandatory, but this requirement is expected to expand as technology moves on.

Reduced vertical separation minima (RVSM) are currently only mandatory in Europe and the Atlantic Ocean areas to support higher traffic densities. "We have made several upgrades and modifications to our MD-11s over recent years," says Tapio Leskinen, MD-11 fleet manager at Finnair. "These include an EGPWS installation in line with Joint Aviation Authorities (JAA) and Federal Aviation Administration (FAA) requirements, a flight management computer (FMC) 921 upgrade, the so-called 'Pegasus modification', and a predictive windshear installation. Finnair has also carried out a replacement modification to the instrument landing system (ILS), installing a multi mode receiver (MMR) with enhanced GPS navigation capability." This last modification is useful looking to the future. The basic form of area navigation requirements (B-RNAV) is mandatory in Europe, with precision (P-RNAV) optional for now, but will be required to fly into major airports in the near future with preferential slots.

The Finnair modification provides support of future P- and required navigation performance (RNP) -RNAV requirements. In addition to these, mode-S transponders are also mandatory, with the elementary and enhanced surveillance becoming mandatory in 2007.

In North America some avionics requirements differ. As with Europe, 8.33kHz frequency spacing and 25kHz frequency spacing are mandated, as are TCAS, EGPWS and Mode-S transponders.

Other modifications

Boeing is also known to be working on changes to system fittings and valves to prevent uncommanded system shutdowns. It has developed modified tail pylon fairings in conjunction with General Electric after several operators reported skin cracks in these areas. Boeing is also experimenting with new coatings to counter unexpectedly high levels of winglet erosion.

There is also the option to add an on-board maintenance terminal (OMT) that provides access to the central maintenance computer messages, together with technical documents and other software applications that can be used on the ground. FedEx has installed a number of these units to its aircraft. **AC**

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MD-11 fuel burn performance

The fuel burn performance of the MD-11 in passenger and freighter configuration is analysed on routes of 3,600-5,800nm.

The MD-11 fleet is dominated by the MD-11ER and MD-11F/CF freighter variant. There are only two engine types offered, from General Electric (GE) and Pratt & Whitney (PW): the CF6-80C2D1F engine from GE; and the PW4462 from PW. GE's is the predominant engine type. There are four maximum take-off weights (MTOWs) for all variants of aircraft, but most aircraft in the fleet have been upgraded to 625,500lbs or 630,500lbs. The fuel burn of the MD-11ER and MD-11F/CF are analysed.

MD-11 variants

The MD-11ER is the passenger version of the aircraft with an MTOW of 630,500lbs for both the CF6-80C2D1F and the PW4462 engine variants. It is predominantly a long-haul aircraft and has a standard fuel capacity of 38,615 US Gallons (USG). Although there are auxiliary tank options for the aircraft, the majority were delivered in standard configuration and have not been upgraded.

Most of the converted and factory freighters have an MTOW of 630,500lbs for both the CF6-80C2D1F and the PW4462 engine variants. The fuel burn difference between the engine variants depends on their fuel burn efficiency. Aircraft with a high fuel burn will require a total higher weight of fuel, which will have an impact on the payload that can be carried when the aircraft is being operated at its MTOW on relatively long routes.

Most MD-11ERs have an MTOW of 625,500lbs or 630,500lbs. The passenger

aircraft have a standard tri-class passenger load of 298.

Fuel burn performance

The fuel burn analyses have been conducted on typical long distance routes for the MD-11. These are representative of both the passenger and freighter routes operated by many of these aircraft.

The routes for the passenger aircraft are Copenhagen (CPH) - Shanghai (PVG), and Madrid (MAD) - Buenos Aires (EZE). The aircraft have been analysed with a full load of 298 passengers, unless the aircraft is payload limited.

The routes used for the freighter aircraft are London Gatwick (LGW) - Nairobi (NBO) and Miami (MIA) - Sao Paulo (GRU), which are typical freight routes.

To illustrate the effects of wind speed and direction, the performance of the aircraft have been analysed for operations in both directions. The differentials on some of these city-pairs have an impact on flight time, available payload and fuel burn.

The tracked distance for the sectors is as follows:

- 4,714nm for CPH-PVG
- 5,486nm for MAD-EZE
- 3,557nm for MIA-GRU
- 3,665nm for LGW-NBO.

Flight plans have been prepared by Navtech, Canada to analyse the fuel burn and payload-carrying performances. The standard weight for each passenger plus baggage has been assumed to be 220lbs.

It is assumed that all the engine configurations incorporate all the performance improvement modifications.

The performance of all aircraft has been analysed with the aircraft operating at long-range cruise, annual 85% winds and a taxi time of 25 minutes.

MD-11ER

The specification weights of the MD-11ER are shown (see table, this page), and are identical for each engine variant. The aircraft is assumed to fly with a full load of 298 passengers in tri-class, giving it a total payload of 65,560lbs. The aircraft has an MTOW of 630,500lbs, an operating empty weight (OEW) of 291,120lbs, and a fuel capacity of 38,615 USG (see table, this page).

The two routes used for the analysis have a tracked distance that is 1,000-2,000nm shorter than the aircraft's 6,500nm range capability with a full passenger payload.

The first route examined for the MD-11ER was CPH-PVG. This has a tailwind of 11 knots, decreasing the tracked distance of 4,714nm to an equivalent still air distance (ESAD) of 4,608nm (see table, page 18).

Flying west, the aircraft experiences a headwind of 49 knots flying towards CPH. This increases the tracked distance to an ESAD of 5,179nm.

The take-off weights (TOW) of the two engine variants differ due to differences in engine fuel burn, resulting in different block fuel for the sector.

The TOW of the PW4462 powered aircraft is 550,725lbs, compared to a TOW of 536,754lbs for the CF6-80C2D1F-powered aircraft (see table, page 18).

The second route examined is MAD-EZE. This is a good example of the long-range routes the aircraft was designed for. The tracked distance of 5,486nm in a westerly direction to EZE faces a headwind of 30 knots. This increases the ESAD to 5,855nm. Going east, to MAD, there is an average 5 knot tailwind that reduces the tracked distance to an ESAD of 5,494nm (see table, page 18).

MD-11F/CF

The freighter aircraft have an OEW of 248,567lbs, and maximum zero fuel weight (MZFW) of 461,300lbs. This gives the aircraft a gross structural payload of 212,733lbs and a fuel standard capacity of 38,615 USG.

The freighter aircraft (F and CF) have been analysed in both directions on MIA-GRU and LGW-NBO as representative of a long-haul freight routes. Both these routes have a tracked distance similar to the freighter's range of about 3,600nm with a maximum payload.

WEIGHT SPECIFICATIONS OF ANALYSED MD-11 VARIANTS

Aircraft type	MTOW lbs	MZFW lbs	OEW lbs	Payload lbs	Seats	Fuel USG	Engine model
MD-11ER	630,500	400,000	291,200	108,880	298	38,615	PW4460/62/CF6-80C2D1F
MD-11F/CF	630,500	461,300	248,567	209,880	N/A	38,615	PW4460/62/CF6-80C2D1F

FUEL BURN PERFORMANCE OF MD-11

City-pair	Aircraft variant	TOW lbs	Engine model	Block fuel USG	Block time	Passenger payload	Fuel USG per passenger	ESAD nm	Wind speed factor
CPH-PVG	MD-11ER	550,725	PW4462	24,642	10:11	298	82.7	4,608	+11
CPH-PVG	MD-11ER	536,754	CF6-80C2D1F	23,226	10:14	298	77.9	4,608	+11
PVG-CPH	MD-11ER	578,925	PW4462	28,403	11:22	298	95.3	5,179	-49
PVG-CPH	MD-11ER	564,056	CF6-80C2D1F	26,728	11:18	298	89.7	5,175	-49
MAD-EZE	MD-11ER	607,979	PW4462	32,663	12:47	298	109.6	5,855	-30
MAD-EZE	MD-11ER	590,211	CF6-80C2D1F	30,703	12:43	298	103.0	5,853	-30
EZE-MAD	MD-11ER	590,598	PW4462	30,224	12:01	298	101.4	5,493	5
EZE-MAD	MD-11ER	574,071	CF6-80C2D1F	28,439	11:58	298	95.4	5,494	5

Source: Navtech

FUEL BURN PERFORMANCE OF MD-11

City-pair	Aircraft variant	TOW lbs	Engine model	Block fuel USG	Block time	Freight payload lbs	Fuel USG per ton-mile	ESAD nm	Wind speed factor
MIA-GRU	MD-11F/CF	630,500	PW4462	22,853	7:59	200,044	0.071	3,595	-5
MIA-GRU	MD-11F/CF	630,500	CF6-80C2D1F	22,080	7:56	207,783	0.066	3,594	-5
GRU-MIA	MD-11F/CF	630,500	PW4462	23,508	8:14	195,693	0.073	3,724	-16
GRU-MIA	MD-11F/CF	630,500	CF6-80C2D1F	22,739	8:11	203,235	0.068	3,723	-16
LGW-NBO	MD-11F/CF	630,500	PW4462	26,362	9:28	175,323	0.090	4,295	-70
LGW-NBO	MD-11F/CF	630,500	CF6-80C2D1F	25,497	9:22	183,701	0.083	4,288	-70
NBO-LGW	MD-11F/CF	630,500	PW4462	22,768	7:58	200,653	0.067	3,582	20
NBO-LGW	MD-11F/CF	630,500	CF6-80C2D1F	22,001	7:55	208,536	0.062	3,583	20

Source: Navtech

En-route winds are not significant for MIA-GRU as this is a North-South route. There is a five-knot headwind in a southerly direction to GRU, increasing the tracked distance to an ESAD of 3,595nm.

North to MIA, the tracked distance increases from 3,600nm to an ESAD of 3,724nm due to a 16-knot headwind.

On LGW-NBO, the winds are significantly different in each direction. South towards NBO the 70-knot headwind adds 630nm making the ESAD 4,295nm. North towards LGW the 20-knot tailwind reduces the ESAD to 3,582nm.

Fuel burn results

The passenger variants of the MD-11 with PW4462 or CF6-80C2D1F engines have different fuel burn performance. The PW4462-powered aircraft burns just over 6% more fuel on the CPH-PVG route than the CF6-80C2-powered aircraft (see table, this page), which is even more noticeable on the westerly routing against a strong headwind. Going in an easterly direction towards PVG, the difference in fuel burn per passenger is 4.75 USG (see table, this page), which is equal to an additional cost of \$9.5-10.5 per seat at current fuel prices. This clearly illustrates the economic advantage of the CF6-80C2D1F-powered aircraft.

Going west towards CPH the difference increases to over 5.6 USG and \$12.4 per seat. This is also a consideration when looking at freighter conversion of the passenger aircraft, as fuel burn differences translate directly into revenue payload differences.

Similar differences in fuel burn between the two engine types are seen on the MAD-EZE route.

The aircraft with both engine types operate at a take-off weight lower than MTOW, but the lower fuel burn of the CF6-80C2-powered aircraft requires a lower total fuel load, which is reflected by a lower take-off weight (see table, this page).

Despite these differences in fuel burn, the aircraft are carrying relatively low payloads and are not limited in the payloads they can carry.

The ESADs of the MIA-GRU and LGW-NBO routes mean that both engine variants of the MD-11F/CF operate at the aircraft's MTOW of 630,500lbs.

The lower fuel burn of the CF6-80C2-powered aircraft is reflected in its superior operating performance at MTOW on these long routes. It can carry close to a maximum payload in both directions on the MIA-GRU route (see table, this page). The northerly direction has an ESAD of 3,724nm, and the aircraft can carry a payload of 207,783lbs, which is only about 5,000lbs

less than maximum. The PW4462-powered aircraft, however, requires a high fuel load and correspondingly has a 9,750lbs lower payload. The CF6-80C2D1F-powered aircraft has a fuel burn of 0.066 USG per available ton-mile (see table, this page). The PW4462 burns 0.071 USG per available ton-mile.

The PW4462-powered aircraft has a similar payload deficit operating in the other direction. Both aircraft carry lower payloads because of the longer ESAD.

LGW-NBO is about 700nm longer than MIA-GRU because of the stronger headwind. This illustrates the effect of lower payload, while again the CF6-80C2 powered aircraft has a superior performance. This aircraft can carry about 8,400lbs more freight, but it is still about 29,000lbs less than a maximum payload. The PW4462-powered aircraft can carry about 37,000lbs less than a maximum payload.

The CF6-80C2-powered aircraft has a similar advantage operating in a northerly direction (see table, this page).

In the northerly direction with a tailwind, the CF6-80C2D1F can carry only about 4,000lbs less than a maximum payload, while the PW4462 has a restriction of about 12,000lbs (see table this page). **AC**

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MD-11 maintenance analysis & budget

Given its size and ability as a high capacity long-haul freighter, the MD-11 has competitive and steady maintenance costs. The aircraft is likely to continue operating for 15-20 years.

Only 200 MD-11s were ever built, but with 195 remaining in operation, the aircraft has become established as an important freighter type. It is also relatively young, ranging in age from six to 16 years old. The MD-11F is also in a class of its own in terms of the payload capacity it offers to freight carriers. There are no direct alternatives, since the 777 factory freighter has a higher payload but also a disproportionately higher financing cost. The MD-11 is therefore a sought-after aircraft, whose popularity is expected to continue for the next 15-20 years.

Only 34 MD-11s remain in passenger configuration, eight of which have already been purchased for conversion to freighter. The remaining 26 are likely to be retired by their operators within the next 10 years, and subsequently converted to freighter. In 1999 UPS stated its original intention to acquire 35 MD-11s for conversion to freighter, and has now acquired 37. This reduces the number of passenger-configured aircraft available to 26 units, which are in high demand.

The MD-11's full maintenance costs are analysed here, including: line and ramp maintenance; base check maintenance; engine repair and overhaul; rotatable repair and management; and heavy component repair and overhauls.

MD-11 in operation

The MD-11 is used as a medium- and long-haul aircraft in both passenger and freight operations. The freighter's range with a maximum payload is 3,700nm, equal to a flight time of eight hours. The route networks of many freight carriers

mean that their average flight cycle (FC) times are in the region of five or six flight hours (FH). Freighters experience high annual rates of utilisation, which in many cases reach 4,500-5,000FH per year. This means that many aircraft accumulate 800-900FC annually. Lufthansa Cargo, for example, has a large global route network and an average FC time of 5.7FH. Its aircraft accumulate about 4,800FH and 830FC per year.

Finnair has a fleet of seven passenger-configured aircraft. These are operated on its long-haul network from Helsinki, which includes New York and various points in the Asia Pacific. The aircraft have an average utilisation of about 5,000FH per year, an average cycle time of about 7.0FH, and generate about 700FC annually.

The MD-11's full maintenance costs are examined for a passenger aircraft completing 5,000FH and 700FC per year, at an average FC time of 7.0FH, and for a freighter aircraft completing 4,500FH and 820FC per year, at an average FC time of 5.7FH.

Line maintenance programme

Like all aircraft types, the MD-11 has pre-flight (PF) checks made before the first flight of each day's operations, a transit (TR) check prior to all other flights during the day, and a daily check. Under maintenance steering group 3 (MSG3) philosophy, a daily check can now be performed up to every 48 hours. Daily checks are usually performed when the aircraft is present for a sufficiently long downtime at its homebase. In the case of long-haul aircraft, like the MD-11, this can be more than once every 24 hours.

After the daily check, the A check is the next highest check in the MD-11's maintenance programme. "The MD-11's original 1A tasks interval in the maintenance review board (MRB) in 1990 was 350FH," explains Tapio Leskinen, manager MD-11 fleet engineering at Finnair Technical Services. "We have been able to escalate this interval several times since 1990 and our current interval for the 1A tasks is 700FH. There are also multiples of A check tasks: 2A tasks with an interval of 1,400FH; 3A tasks at 2,100FH; 4A tasks with a 2,800FH interval; and 6A items with a 4,200FH interval. The full cycle of A check tasks is therefore completed at the A12 check, which has a maximum interval of 8,400FH."

The MD-11's maintenance planning document (MPD) now has an interval of 600FH for the basic 1A check items. There are also tasks that have multiples of this interval: the 2A, 3A, 4A and 6A tasks. The 6A tasks therefore have an MPD interval of 3,600FH. The A12 check and the A check cycle would have an interval of 7,200FH.



The MD-11 is operated as a long-haul aircraft both in passenger and freight operations. The aircraft is capable of intensive use, achieving annual utilisations of up to 5,000FH per year.



Other experienced operators have a longer basic 1A interval. “We operate a fleet of 19 aircraft. Some of these are factory-built freighters, while other older aircraft are converted passenger aircraft,” explains Thomas Streychez, manager of MD-11 system engineering at Lufthansa Technik. “We have escalated our 1A interval to 720FH during the course of our operation.” This means that the maximum possible interval for the A check cycle is 8,640FH, although in all operators’ cases the actual interval achieved between A checks is 80-85% of the MPD interval. Most MD-11 operators have an A check interval of 600FH, so they would be performing an A check every 450FH, and completing the A check cycle after 5,400FH.

Base maintenance programme

The MD-11’s MPD is based on MSG3 philosophy. The ageing aircraft programme items that were added on to the DC-10’s maintenance programme are therefore an integral part of the MD-11’s programme.

Leskinen says that the MD-11’s MRB initially had an interval of 4,200FH and 15 months, whichever is reached first, when it began operations in 1990. This meant that the aircraft would be limited to an annual utilisation of 3,360FH if both intervals were to be fully utilised.

Leskinen explains that Finnair, which is one of the first and most experienced MD-11 operators, has managed to have its own C check intervals extended up to 7,500FH and 18 months.

The MD-11’s MPD has a C check interval of 6,000FH and 15 months, which allows a maximum of 4,800FH per

year if both intervals are to be reached at the same time. There are multiple tasks, the heaviest of which are the 4C items, and the base check cycle culminates with the C4 check. The base check cycle therefore comprises the C1, C2, C3 and C4 checks.

The C4 check, which is sometimes referred to as the D check, is the last and heaviest check in the base maintenance cycle. This has an interval of 24,000FH and 60 months. The base maintenance cycle is therefore completed every five years, or about every four years considering most operators’ utilisations of check intervals.

“Our C4 check interval is one of the longest in the global fleet: 30,000FH and 72 months,” says Leskinen. “This means that our full base check cycle now has an interval of up to six years, but the shorter intervals that the aircraft originally had mean that the first C4 checks were performed at an age of about five years. Our oldest aircraft will have their third C4/C12 check in the next year.”

Lufthansa Cargo has a 1C task interval of 6,000FH and 15 months, which allows up to 4,800FH per year if both intervals are to be completely utilised. The airline’s 4C check interval is 24,000FH and 60 months.

The length of the base check cycle, and probable check interval utilisation, mean that the oldest MD-11s will now have been through, or will just be approaching, their third C4 check. The youngest and last MD-11s were built in 1999 and 2000. These aircraft will have been through their first C4 check, and will be approaching their second in two to three years’ time. Most of the MD-11 fleet is therefore mature.

The MD-11’s base maintenance programme is simple, with a series of four checks culminating in a heavy check. The C4 check is used to refurbish the aircraft interior and overall usually is a large workscope. The aircraft has so far demonstrated a low non-routine ratio.

Base check contents

C checks include several elements that are present in base checks for all aircraft. The first of these comprises routine inspections for corrosion and other items that were treated as separate ageing aircraft items. These routine inspections are accompanied by non-routine rectifications and defects that arise out of the routine inspections.

The MD-11 also has some structural sampling tasks that are required on a percentage of an operator’s fleet.

As with all types of aircraft, there are also out-of-phase tasks. These are items that have fixed or hard-time maintenance intervals that do not coincide with A or C check intervals, but are normally scheduled into one of these checks to avoid additional maintenance downtime to take care of these issues. Janne Tarvainen, assistant vice president component department at Finnair Technical Services, explains that of the MD-11’s 1,300 rotatable components, about 10% are hard-time components. These are safety items, batteries and items such as the integrated drive generator and the radio cooling fan.

The C4 or D check is the check during which most operators will carry out the larger items of interior refurbishment. This can include the refurbishment of galleys and toilets, overhead bins and passenger service units, sidewall panels and carpets, and seats. Seats and carpeting may also be refurbished on an on-condition basis and taken care of during the lighter C checks.

“Freighter aircraft will use up to 10,000MH fewer for a full C4 check package, because freighter aircraft have almost no interior refurbishment requirements,” explains Leskinen. “Passenger aircraft have many additional items that consume a lot of MH for refurbishment.”

The C4 check has a downtime of about four weeks, which also presents an opportunity to strip and repaint the aircraft. Leskinen explains that this consumes 2,500MH and \$50,000 in materials.

Service bulletins (SBs), modifications and airworthiness directives (ADs), must also be considered. These can account for several hundred or thousand MH of a complete C or C4 check workpackage.

"We have made several upgrades and modifications to our MD-11s over recent years, which include a flight management computer (FMC) 921 upgrade, an enhanced ground proximity warning system (EGPWS) installation, a predictive windshear installation and various other avionic modifications. We also get SBs issued by Boeing, some of which are individual structural inspections. There are also several system and avionic upgrades," says Leskinen.

"The MD-11 has also had three major ADs issued against it, which have all been dealt with across the fleet, but which have added a large number of MH to complete the modification," continues Leskinen. "The first of these was the engine pylon upper spar modification, which used about 900MH per engine pylon. The second was AD 2000-11-02, relating to the installation of a new cabin insulation blanket to provide better protection against fire. This has been completed on all MD-11s, but used about 10,000MH per aircraft to complete. The third major AD involved inspection and treatment of the horizontal stabiliser barrel nuts, which used about 1,500MH per aircraft. This has now been recently completed on all MD-11s."

Brian O'Meagher, components monitoring centre manager VEM Maintenance & Engineering, explains

that typical SBs used in C checks are the torque shaft bearing inspections on the passenger doors, while SB MD11-53-066 requires an inspection on the left and right main landing gear doors that takes about 40MH to complete.

Line maintenance inputs

TR and PF checks were originally carried out by flight engineers, but were handed over to line mechanics with the advent of aircraft with two-man flightdecks. Many airlines now have these checks performed by the flightcrew, which in most cases require less than 1MH to complete and use just a few dollars of materials and consumables.

A conservative budget would be for 2MH used in each TR and PF check, together with about \$20 for materials and consumables. A line maintenance labour rate of \$70 per FH would take the cost for each check to about \$160.

Finnair Technical Services estimates that daily checks use an average of 6MH and \$80 in materials and consumables. A labour rate of \$70 per MH would take the total cost of a daily check to \$500.

The rate per FH for line and light checks can be analysed in terms of total costs for all these checks performed during the course of a year. The number of daily checks depends on the style of

operation. In the case of aircraft operating one return flight per day, as with Finnair, a daily check would be performed on most days, or every second flight when the aircraft was at the homebase. This would total about 350 daily checks. The pattern of operation between different freight carriers varies between short-, medium- and long-haul. Either way, most aircraft will require a similar number of daily checks in the year, which will also be close to about 350. The average cost of \$500 for each daily check will take the annual cost to \$175,000.

The number of PF and TR checks will be equal to the number of annual FC, less the number of daily checks. In the case of passenger aircraft used on long-haul missions this will be close to 350. This will take the annual cost of TR and PF checks to about \$56,000.

Freighter aircraft operating at a shorter average FC time and generating about 850FC per year, will require 500 TR and PF checks each year, and will have an annual cost of \$80,000.

Passenger aircraft will consume a total of \$230,000 per year for TR, PF and daily checks. Amortised over a utilisation of 5,000FH, this will be equal to \$46 per FH (see table, page 30).

Freighter aircraft will consume a total of about \$255,000 per year for line and

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The C4 check for passenger-configured MD-11s consumes about 43,000MH, while its freighter counterpart will consume about 10,000MH less. This is one main factor that makes the MD-11F/CF competitive against the ageing 747-200SF, which will consume 90,000-100,000MH in its D check.

light checks. Amortised over a utilisation of 4,500FH, this will be equal to a rate of \$57 per FH (see table, page 30).

A check inputs

A checks are performed in block checks by most operators, and consequently vary in size. "We use an average of 440MH for our A checks," explains Streyczek. "This is for the whole A check package, and includes 120MH used for routine tasks. The package will also include rectifications arising from the routine tasks, clearing of deferred defects, engineering orders (EOs) and modifications, cleaning and some out-of-phase tasks."

Finnair uses a block check system, and aims to clear all defects during the A check to maintain a reliable operation. "We basically use each of our aircraft on one return flight per day, and we do not stock an inventory of rotables at outstations. We therefore like to clear technical defects not long after they occur," explains Erkki Lehtonen, assistant vice president aircraft heavy maintenance at Finnair Technical Services. "We get about one defect for each flight. Some of these can be dealt with between flights, and others have to be dealt with at stops because they are no-go items. Deferred items can be cleared during the daily and A checks. The auxiliary power unit (APU) and in-flight entertainment (IFE) system present the most regular problems.

"The A check therefore varies in size and content," continues Lehtonen. "We use 400-500MH for an A check, since they vary in size and content. The main factor causing variation in size is the clearing of technical defects."

The consumables and materials used vary, and mainly include oils, lubricants and small items. The average for an A check will be in the region of \$11,000. A labour rate of \$70 per MH would take the total A check cost to about \$46,000. This would be equal to a rate of \$105 per FH (see table, page 30), if the average interval between A checks is 450MH.

Base check inputs

The elements that comprise the C check packages have been described. The lower C checks (the C1, C2 and C3



checks) have a downtime of about five days, and may include some interior work on the seat covers and carpets. "The C6 checks and the C3 check in the second base check cycle actually have some structural inspections with an initial threshold of 90 months and a repeat interval of 60 months," explains Streyczek. "This means that they are repeated every fourth C check in most cases. They are therefore repeated at each third C check in the base maintenance cycle, and so will occur again at the C10, C14 and C18 check, and every fourth C check thereafter. This means that the routine content of the C checks rises during the second base check cycle.

"We had subcontracted all lighter C checks, but started on the C1/C2/C3/C5/C7 checks at our Frankfurt base earlier in 2006," continues Streyczek. "The routine inspections, rectifications and allowance for findings of these lighter C checks consume about 2,700MH, while another 600MH are used for EOs, SBs, modifications and clearing deferred items. This is a sub-total of about 3,300MH. An A check is also included in the C check, as are major modifications and major component changes. This will take the total for these lighter C checks to 4,000-4,500MH for freighter aircraft."

Leskinen at Finnair says its MD-11s use an average of 8,300MH for lighter C checks, depending on the workpackage. This includes about 3,200MH for routine inspections, 2,300MH for non-routine rectifications, 2,200MH for EOs and modifications and another 600MH for additional work. These aircraft are passenger-configured and will require a higher MH input because of interior-related items. The accompanying

consumption of materials and consumables totals in the region of \$167,000, and 10% of this is for interior work.

The C4 check is used by most operators to complete interior refurbishment and stripping and repainting, and will comprise a large workpackage when all items are considered. Leskinen says that these checks take about four weeks to complete, although they will take longer when items such as the insulation blanket modification is being performed. "The C4 checks on passenger-configured aircraft consume an average of about 43,000MH for the full package, which includes interior refurbishment and stripping and repainting," says Leskinen.

The full package includes about 15,400MH for routine inspections, 13,300MH for non-routine rectifications, another 10,500MH for EOs and modifications and about 3,500MH for additional work. The total includes about 2,500MH for stripping and repainting, as well as about 6,500MH for interior refurbishment. Overall, freighter aircraft use about 10,000MH less for a check of about the same workpackage; the difference being explained by the lack of interior-related work.

"This number of MH is for a C8 check, which is the second heavy check at 10 to 11 years of age. This does not include a large modification, such as changing the insulation blankets on the aircraft. The associated cost of materials and consumables for the whole workpackage is in the region of \$1.18 million, including all items for interior refurbishment," continues Leskinen.

Streyczek points out that freighter aircraft do, however, have a cargo

MD-11 FAMILY HEAVY COMPONENT MAINTENANCE COSTS

FH & FC per year	
Average FC time of FH	
Number of main & nose wheels	10 + 2
Tyre retread interval-FC	180/230
Tyre retread cost-\$	500/400
Number of retreads	4
New main & nose tyres-\$	1,950/1,200
\$/FC retread & replace tyres	42
Wheel inspection interval-FC	1,400
Main & nose wheel inspection cost-\$	2,600
\$/FC wheel inspection	19
Number of brakes	10
Brake repair interval-FC	1,400
Brake repair cost-\$	70,000
\$/FC brake repair cost	500
Landing gear interval-FC	4,500-5,500
Landing gear exchange & repair fee-\$	750,000
\$/FC landing gear overhaul	140-167
Thrust reverser repair interval-FC	3,500
Exchange & repair fee-\$/unit	200,000
\$/FC thrust reverser overhaul	175
APU hours shop visit interval	2,500
APU hours per aircraft FC	2.0
APU shop visit cost-\$	300,000
\$/FC APU shop visit	240
Total-\$/FC	1,116/1,145
Total-\$/FH passenger aircraft @ 7.0FH per FC	160
Total-\$/FH passenger aircraft @ 5.5FH per FC	210

loading system that has to be examined as it is subject to damage during operation. This therefore requires some refurbishment work. "We remove and refurbish the cargo loading system every C check," says Streyczek. "The power drive units are particularly important."

Summary base checks

Mature passenger-configured aircraft consume a total of 68,000MH and \$1.68 million of materials and consumables in a full base check cycle of four checks. At a labour rate of \$50 per MH this will be a total cost of \$5.1 million.

In many cases this will be over a maximum interval of 60 months, although typical levels of check interval utilisation mean that the actual full interval is more likely to be 50-53 months. An aircraft operating at an annual utilisation rate of 5,000FH per year will accumulate 21,000-22,000FH between checks. The reserve for these C

checks will therefore be in the region of \$230-245 per FH (*see table, page 30*). This might be reduced to about \$215 per FH for operators that have extended check intervals. These rates are for mature aircraft going up to their second C4 check, however, and MH and material expenditure are likely to rise as the aircraft pass through their third and fourth base check cycles.

Freighter aircraft will consume a total of about 48,000MH for the full base check, and in the region of \$1.15 million in materials and consumables. A labour rate of \$50 per MH would take this to a total of \$3.55 million.

This will be amortised over the same calendar interval as the passenger aircraft: 50-53 months in most cases considering usual rates of check interval utilisation. On the basis that freighter aircraft accumulate about 4,500FH per year, the reserve for base checks over an interval of 19,000FH would be \$180-190 per FH (*see table, page 30*).

Rotable components

Rotable components and heavy rotable components must be considered separately. Heavy components include landing gear, thrust reversers, wheels and brakes, and the APU. These are considered separately from all other rotatables.

Besides these four types of heavy components, up to 1,800 different rotable components are installed on the MD-11. The number can be as low as 1,300, but will vary considerably according to aircraft configuration. "There are, actually, rotable and repairable components, and so the number of total components on the aircraft depends on the definition used," explains Tarvainen at Finnair Technical Services. "We track about 650 rotable components with a computerised system for the purpose of monitoring aircraft configuration, maintenance planning, and managing inventory.

"The MD-11 is a relatively modern aircraft and about 90% of its rotable components are maintained on an on-condition basis, while the rest have hard-time maintenance intervals," continues Tarvainen. This implies that only about 120 rotatables are maintained on a hard-time basis. Tarvainen explains that these are items which have life limited parts (LLPs), and so their lives and utilisations need to be monitored. The other components with hard-time maintenance intervals are safety-related components such as gas bottles, escape slides, and the standby horizon on the flightdeck. Other hard-time components include batteries, the integrated drive generator (IDG) and radio cooling fan. The number of hard-time rotatables is far smaller than on older generation aircraft, such as the DC-10. "The higher percentage of components that are maintained on an on-condition basis is partly explained by the increased reliability of the components," explains Tarvainen. "The reliability of on-condition components, such as the engine control unit, has been monitored and soft times for recommended removal have been established. Soft times can of course be exceeded when reliability is good."

Finnair Technical Services is one supplier of rotable support packages to MD-11 operators. "These packages are considered separately to heavy components," says Tarvainen. "The package includes the management of the parts such as logistics, repairing and tracking the components. We subcontract the repair of many of the parts, since we do not have repair capability for all 1,300 or so rotatables on the aircraft. We repair about 70% of the volume in-house.

"There are three elements to this type of rotable support package," continues Tarvainen. "First, the airline is supplied

with a homebase stock to which it has immediate access. These are components that are critical to the operation of the aircraft, and also have a high failure rate. The second part of the package involves providing the airline with access to a pool of the remaining rotatables at our own stock. These often form the majority of rotatables an airline will use. The third part of the package is providing logistics services relating to both groups of components. This includes transporting failed and serviceable units, testing and repairing removed components, and providing all relevant airworthiness and serviceable documents.”

The size of the homebase stock is affected by several factors, including: aircraft operation and utilisation; fleet size; age and condition of the aircraft; the component configuration of the aircraft; whether the aircraft is in passenger or freighter configuration; and the insurance level required by the operator. Tarvainen estimates that the list price of homebase stock for a fleet of five aircraft is \$4.5-6.0 million, while another \$2.5 million of stock is required for a fleet of 10 aircraft. The associated lease rate for these components is \$35,000-45,000 per month for a fleet of five aircraft, and \$65,000-75,000 per month for 10 aircraft. This comes to \$85,000-110,000 per aircraft per year, equating to \$17-22 per FH for an aircraft accumulating

5,000FH annually. The rate is reduced to \$16-18 per FH for a larger fleet of 10 aircraft.

Fees for the access pool are \$30,000-35,000 per month for a fleet of five aircraft, and \$50,000 per month for 10 aircraft. This translates to a rate of \$12-15 per FH per aircraft, depending on fleet size.

“The third element for repair and management is about \$190-260 per FH for passenger aircraft, and about \$160 per FH for freighter aircraft,” says Tarvainen.

These three elements total \$220-300 per FH for passenger aircraft, and \$190-210 per FH for freighter aircraft (see table, page 30).

Heavy components

As described, these comprise four main component types: wheels and brakes, landing gear, thrust reversers and the APU.

Removals of wheels and brakes are first triggered by tyre wear and a need for remoulding. There are 10 main wheels. Finnair reports an average interval of 230FC for remoulds, and 180FC for nose tyre remoulds. Main tyres have a life limit of 1,200FC and so are remoulded up to four times before being replaced at the fifth removal, while nose tyres have a limit of 800FC and are remoulded up to

three times.

Remoulds cost about \$500 for main tyres and \$400 for nose tyres. New main tyres cost \$1,950 each and new nose tyres \$1,200 each. The life cycle remoulding and replacement cost for a complete shipset of 12 tyres is about \$22,000, which is equal to a reserve of \$42 per FC (see table, page 24).

Leskinen at Finnair explains that wheel rim inspections are made every sixth tyre change, and cost \$2,600 per wheel. This has a shipset cost of \$26,000, which, amortised over an interval of 1,400FC has a reserve of \$19 per FC (see table, page 24).

The MD-11 has carbon brakes and Leskinen explains that these have an average repair interval of 1,400FC, and associated average repair cost of \$70,000, including the cost of a new heat pack. This is equal to a reserve rate of \$500 per FC for all 10 brake units (see table, page 24).

The landing gear has a hard-time removal interval of eight years and 7,500FC, whichever is reached first. Most aircraft accumulate in the region of 700FC per year, and so about 5,500FC over an eight-year removal interval.

Most airlines choose to have an exchange that includes an element for the capital cost of the landing gear, the repair and overhaul cost, and any additional costs that may be incurred. A market rate



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