

# GE90 family maintenance costs

**Removal intervals, shop visit workscope patterns & costs, LLP lives and reserves provide a guide to maintenance reserves for the GE90 Standard and GE90 Growth engines at typical rates of operation & utilisation.**

**T**he GE90-powered fleet of 777s is the largest sub-fleet of 777s in operation. There are about 650 aircraft in service with GE90 engines, and a further 330 aircraft on firm order. The majority of aircraft are operated on long-haul services.

The fleet can be sub-divided between the smaller variants with the 123-inch fan, and those with the larger 128-inch fan. Most engines in the 123-inch-fan fleet are mature in age and maintenance terms. The oldest 128-inch fan engines are 10 years old, although the engines did not come into service in significant numbers until 2004. The oldest 128-inch fan engines will have been through at least two major shop visits. The majority of engines in the fleet will have either been through only one shop visit, or are still in their first removal interval.

The maintenance reserves of both main variants operated at different rates of utilisation and at different engine flight hour (EFH) to engine flight cycle (EFC) ratios are examined.

## GE90 in operation

There are about 170 aircraft in active service with 123-inch-fan engines. These are eight to 17 years old; with most aircraft being delivered up to 2007, although the final two were delivered in 2010. The fleet can be sub-divided into four fleets of the -76B, -85B, -90B and -94B. These are rated at 76,000lbs to 93,700lbs thrust (*see GE90 Specifications, page 4*). The -76B and -85B fleets are small, with just 32 aircraft in total. These power early-built 777-200s and -200ERs, and are operated mainly by British Airways (BA) and a smaller number by China Southern Airlines. The BA fleet is operated on medium-haul services to the Middle East and some East Coast destinations in the US. China Southern operates its aircraft on Chinese domestic services.

The higher-rated -90B and -94B fleets account for most of the 123-inch-fan fleet, with 137 aircraft in operation. The

biggest operators are Air France, China Southern, Saudia, United Airlines (which are ex-Continental Airlines aircraft), Alitalia, Japan Airlines (JAL), KLM, Pakistan International (PIA) and Vietnam Airlines.

The -90B and -95B fleets are both operated at similar rates of utilisation and EFH:EFC ratios. A minority of the aircraft are operated on medium-haul operations with annual utilisations of 3,000-3,500 flight hours (FH). These are aircraft operated by Saudia, PIA and Vietnam Airlines.

Most aircraft, however, are operated on long-haul networks, and have annual utilisations of up to 5,500FH. EFH:EFC ratios are 8-11:1. Some airlines operate their aircraft on ultra-long-haul operations. These include Air France, Alitalia and United/Continental.

There are almost 480 active aircraft with 128-inch-fan engines. The oldest of these is 10 years, while there are more than 300 aircraft on firm order. The 128-inch-fan fleet can be sub-divided into three sub-fleets. There is the GE90-110B, rated at 110,000lbs thrust, which powers two of the sub-fleets. There are two sub-fleets of 54 777-200LRs and 65 777-200Fs in active service.

The third sub-fleet is the 777-300ER powered by the GE90-115B, rated at 115,000lbs thrust. This has almost 360 active aircraft, and has another 263 on firm order, and so is the largest sub-fleet of GE90-powered aircraft.

The 777-200LR is operated by Air Canada, Air India, Delta, Emirates, Ethiopian Airlines, PIA and Qatar Airways. This is a specialist ultra-long-haul aircraft, although Air India and Ethiopian Airlines operate their fleets on long-haul services of more typical flight lengths. All other operators utilise the 777-200LR on ultra-long-haul services. These have EFH:EFC ratios of 9-11FH, and annual utilisations of 4,500-5,500FH.

The 777-200F fleet includes main fleets operated by Aero Logic, China Cargo Airlines, China Southern,

Emirates, FedEx, LAN Cargo, Qatar Airways, Southern Air and TNT Airways. As with all freight operations, the EFH:EFC ratios of the 777-200F are shorter than the -200LR, even though the aircraft has almost identical maximum take-off weights (MTOWs) and fuel capacities. Most aircraft are operated at flight cycle (FC) times of 4.5-6.5FH, and have annual utilisations in the region of 4,500FH.

The 777-300ER fleet also operates with a mix of average FC times. Most aircraft are utilised for long-haul missions, but fleets operated by Saudia, Emirates, EVA Air and Philippine Airlines are used on medium and shorter long-haul operations of 2.5-7.0FH.

Most other operators use their aircraft on long-haul missions of 7.0-13.5FH. Virgin Australia operates the longest FC times, with aircraft being deployed from Sydney to Los Angeles and Johannesburg. These aircraft achieve annual utilisations of 4,700FH and 350FC.

Turkish Airlines (THY) operates a fleet of 12 aircraft to destinations in North America, Sao Paulo, Hong Kong, Beijing and Singapore and generate about 5,300FH and 570FC per year. Singapore Airlines achieves some of the highest rates of utilisation at about 5,500FH and 650-700FC per year.

## Maintenance management

While a portion of engines are managed under total care style programmes, the removal intervals and workscope of engines generally have to be managed around the life limits of life limited parts (LLPs) and shop-visit workscope.

The GE90 has four main modules. These are the fan and low pressure compressor (LPC), high pressure compressor (HPC), high pressure turbine (HPT) and combustor, and low pressure turbine (LPT). There are a total of 22 rotating LLPs in these four modules in the 123-inch-fan engine (*see table, page 12*), and 23 rotating LLPs in the 128-inch-fan engine (*see table, page 12*).

The target lives for these rotating LLPs are 20,000EFC in the fan LPC and HPC, and low pressure turbine in the 123-inch-fan engine. The rotating parts in the HPT have target lives of 15,000EFC.

The target life of all rotating parts is 15,000EFC in the 128-inch fan engine.

In addition to these, fan blades are also classified as LLPs, and these have lives of 30,000EFC in both main variants (*see table, page 12*). Given the rates of utilisation of most aircraft and engines, these parts are unlikely to require replacing due to life limit expiry in their operational lifetime. They are, however, susceptible to foreign object damage

## GE90 FAMILY LIFE LIMITED PARTS

Engine variant	GE90 Std		GE90-110B/-115B	
	Certified life EFC	Target life EFC	Certified life EFC	Target life EFC
<b>Fan/LPC</b>				
Fan blades	30,000	30,000	30,000	30,000
Fan rotor disk	20,000	20,000	15,000	15,000
Booster spool	20,000	20,000	15,000	15,000
Forward fan shaft	20,000	20,000	13,100	15,000
Mid fan shaft	10,000	20,000	9,400	15,000
<b>HPC</b>				
HPC forward shaft			15,000	15,000
Stage 1 disk	19,000-19,100	20,000	15,000	15,000
Stage 2-6 spool	11,000-15,000	20,000		
Stage 2-5 spool			8,800	15,000
Stage 6 disk			8,500	15,000
HPC impeller			15,000	15,000
Stage 7 disk	14,000-15,800	20,000		
HPC 7-9 spool			7,700	15,000
Stage 8-10 spool	9,800-16,700	20,000		
Tube supporter ring	20,000	20,000		
CDP seal	18,000-18,600	20,000	12,400	15,000
<b>HPT</b>				
Forward seal	15,000	15,000	9,000	15,000
Stage 1 disk	9,500	15,000	8,100	15,000
Interstage seal	3,500	15,000	11,400	15,000
Stage 2 disk	14,200	15,000	8,600	15,000
Aft seal	15,000	15,000	15,000	15,000
<b>LPT</b>				
Stage 1 disk	19,800	20,000	15,000	15,000
Stage 2 disk	12,300	20,000	13,800	15,000
Stage 3 disk	14,900	20,000	15,000	15,000
Stage 4 disk	14,600	20,000	8,700	15,000
Stage 5 disk	13,400	20,000	15,000	15,000
Stage 6 disk	20,000	20,000	10,800	15,000
Cone shaft	15,000	20,000	15,000	15,000
<b>Cases</b>				
HPC forward case			25,380	40,000
HPC extension case			7,430	40,000
HPT case	13,370	40,000	6,780	40,000
Combustion case			17,700	40,000
Turbine center frame			14,300	40,000
LPT case	19,800	40,000	24,800	40,000

(FOD) and other types of damage, and so may require replacing for these reasons.

A third group of parts are also classified as LLPs. The high pressure (HP) turbine case and the low pressure (LP) turbine case in the 123-inch engine have target lives of 40,000EFC. The actual certified lives of these parts are 13,370EFC and 17,900EFC (see table, this page). The engines will therefore have to be managed around these life limits.

In the 128-inch-fan engine, the HPC forward case, the HPC extension case, the combustion case, the HPT case, the turbine centre frame, and the LPT case are all classified as LLPs. These six parts all have target life limits of 40,000EFC.

The actual life limits of two of these parts with the shortest lives are 7,430EFC and 6,780EFC. The lives of the other four parts are 14,300EFC to 25,380EFC (see table, this page).

The lives of the rotating LLPs have an impact on engine management. Not only do their target lives of 15,000EFC and 20,000EFC mean that they are likely to need replacement during the operational life of the aircraft, but many of the parts have actual certified lives that are shorter than these target limits. This means that they are likely to require replacement during the aircraft's operational lifetime, so the relatively short life of some rotating LLPs means that engine removals

and shop-visit worksopes will have to be managed around the life limits of these parts.

The actual certified life limit of rotating LLPs for each part has varied since the engine's entry into service. This is first because there are several part numbers for each rotating part. Different part numbers are used on the production line during the manufacture of each engine variant, and a part number with a longer life limit will replace an earlier part number used in earlier-build engines. The removal of different engines of the same variant may therefore have to be managed differently because of their varying LLP life limits.

Another factor is that the life limit of particular part numbers has been increased by General Electric (GE) during their operational life due to operational experience. This may or may not come before the original life limit is reached.

Each variant had one or two rotating LLPs with short life limits. The GE90-76B, for example, had a life limit of just 3,500EFC for its HPT interstage seal in 2008. The same part in the -85B, -90B and -92B variants of the 123-inch-fan engine and the GE90-110B had the same life limit.

The life limit of this part has been increased to 15,000EFC in the 123-inch-fan engine, and to 11,400EFC in the 128-inch fan engine.

Other rotating LLPs in the 123-inch-fan engine had lives of 9,500-11,000EFC in 2008. These have been increased to 15,000-20,000EFC.

This highlights the problem of managing engine removal when LLPs with restricted lives force removals and are replaced with parts with longer lives.

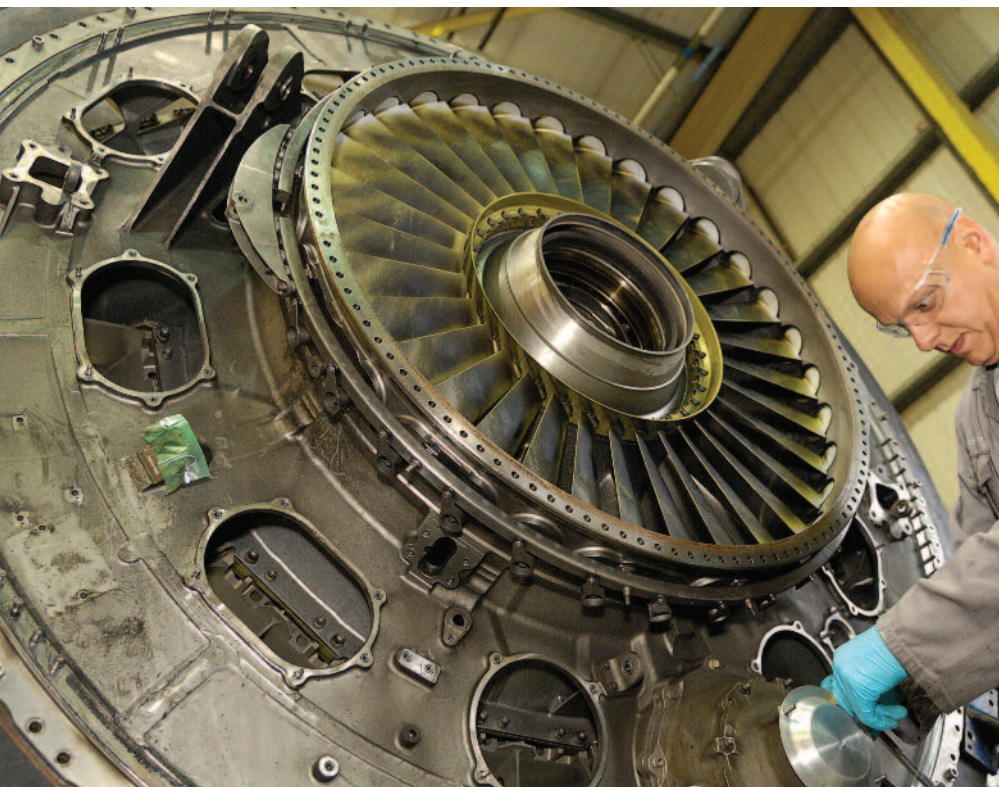
### GE90-76B/-85B/-90B/-92B

As described, the majority of 777-200s are utilised by their operators on long-haul operations. The engines on most of these aircraft operate on average EFC times of 6.0-8.0EFH. These engines have achieved typical first planned removal intervals of 16,000-20,000EFH, equal to 2,000-2,700EFC. At typical rates of aircraft utilisation, this is equal to three-and-a-half to five years of operation.

These intervals would not have been limited by the 3,500EFC limit of the HPT interstage seal. This LLP would have to be removed at the first shop visit, however, to prevent limiting the second removal interval.

### Engines at 2.8-4.5EFH

Engines operating at shorter EFC times of about 3.0EFH were capable of longer EFC intervals to the first shop visit, but these would have been limited to 3,500EFC if the interstage seal with a



life of this limit had been installed on the engine. This is equal to 12,500-14,000EFH, depending on each EFH:EFC ratio.

The same engines, which are mainly the lower rated -76B and -85B, have demonstrated an ability to achieve second removal intervals of 4,500-5,500EFC; the longer interval being possible because of the absence of a LLP limiter. The engine would have been capable of first intervals of 5,000EFC or more. The total time at the second removal would be 8,000-9,000EFC. Removal would ultimately have been forced by a second LLP limit of 9,500EFC by the HPT first stage disk, and also at 9,800EFC by the HPC 8-10 spool.

Aircraft operating at 2.8-4.5EFH per EFC have reached maturity and removal intervals have settled to mature intervals that are close to the second removal intervals of 4,500-5,000EFC. This is equal to 16,000-20,000EFH. The actual interval will depend on the operating environment. Some Saudia aircraft operate in high ambient temperatures, so shorter intervals can therefore be expected.

The earlier-built GE90s had other limited LLPs with lives of 12,300EFC, 13,400EFC, 14,200EFC and 14,600EFC (see table, page 12).

Paolo Lironi, executive director leasing at SGI Aviation explains that at maturity, the engine can broadly follow a shop-visit workscope pattern of a core refurbishment and minor work on the LPT, followed by a broader workscope that will include the fan and LP modules. The engine can therefore be expected to approximately follow an alternating shop

visit pattern.

The pattern of workscope can differ, however, depending on removal intervals. The first shop visit can be a full core workscope, while the second can include a light workscope on the LPT. The fan and LPC module may not be included until the third shop visit, and a full LPT workscope would not be included until the fourth shop visit.

Estimates for the first shop visit are \$5.5-6.5 million, and \$6.5-7.5 million for the second. The cost for the third shop visit is similar to the second shop visit. This would result in a total cost over the three shop visits of \$18.5-21.5 million for a mature engine. Amortised over the total mature interval of 48,000-66,000EFH up to the third shop visit, this would result in reserves for shop visits of \$325-285 per EFH (see table, page 15).

This is the reserve for planned removals, but reserves for unscheduled or unplanned removals must also be budgeted for. "The rate of unscheduled removals for shop visits is 0.05 per 1,000EFH," says Lironi. This is equal to about 20,000EFH. The implication is that an unscheduled visit will come due at an interval about equal to scheduled removals.

Reserves for LLPs also have to be considered. This raises the issue of how much operators can be compensated for the restricted lives of LLPs with lives shorter than the target lives. That is, compensation could come in the form of a replacement part being supplied at a pro-rated price so that LLP reserves are the same as would be experienced for parts with unrestricted lives.

If compensation is possible, then LLP

*While both main GE90 variants have LLPs that can limit planned removal intervals, engines can generally remain on-wing for extended periods, and without removal being forced by loss of EGT margin.*

reserves could be close to the cost of a shipset of parts amortised over the full life limit. The LLPs have to be considered in three main groups. The 17 rotating parts with target lives of 20,000EFC have a list price of \$4.6 million. The five rotating parts with target lives of 15,000EFC have a list price of \$1.8 million. The 22 fan blades, which have a unit cost of \$95,000, have a life limit of 30,000EFC. The fourth group is the two cases that have a list price of just over \$1 million and have a target life of 40,000EFC. The four groups have a total list price of \$9.5 million. When amortised over their full target lives they have a reserve of \$445 per EFC. This is equal to \$100-160 per EFH when amortised over the EFC times of 2.8-4.5EFH (see table, page 15). Given that aircraft operating at these FH:FC ratios will accumulate about 1,000EFC per year, the fan blades and cases will need replacing within 10-14 years.

The total reserve for these two main elements is \$425-545 per EFH (see table, page 15).

Additional consideration has to be given here for the effect of unscheduled removals and shop visits. These can either add to the number of planned shop visits an additional cost that has to be amortised over the same interval, or reduce the total interval of three or four consecutive shop visits.

A further issue is the cost of repair and reserves for the engine quick engine change (QEC) kit. An allowance of \$40-50 per EFH should be made for this.

### Engines at 6.0EFH

The first planned removal intervals of engines operated at 6.0EFH per EFC are 2,700-3,000EFC. This is equal to 16,000-18,000ERH; so a longer EFH interval than engines operated on shorter EFC times. The life limit of the HPT interstage seal at 3,500EFC would clearly prevent a longer interval. The removal of this part would allow an unrestricted second removal interval, since the rotating LLP with the second shortest interval was 9,000EFC in earlier-built engines.

Engines operated at this moderate EFH:EFC ratio achieved second and subsequent removal intervals of 2,200-2,700EFC; which is equal to 13,000-

16,000EFH. Total interval to the second removal will therefore have been 29,000-34,000EFH, equal to six years of operation. While the second and third removal intervals would not have been affected by any LLP life limits, the fourth removal interval may have come close to being compromised by parts with lives of 9,500EFC. These parts, and others with life limits of up to 12,000EFC would have to be removed at the third shop visit to prevent limiting the fourth removal interval. Given that one of these parts was the fan mid-shaft, limited at 10,000EFC, and others are in the HP modules, most of the engine would require a full disassembly. Depending on condition and findings, the engine may require a full disassembly at the second shop visit. The parts with lives of up to 11,000-11,500EFC would then probably have had to be replaced at this stage to prevent forcing a full disassembly of the engine again prematurely at the third or fourth shop visit.

The majority of GE90 123-inch fan engines will have been through their first two planned shop visits. Only a small number of engines manufactured in 2007 and 2010 will have been through one planned shop visit.

Most engines will therefore be at maintenance maturity. The engines will conform to a fairly steady shop visit workscope pattern. This can be where the fan and LP modules can be worked on every second interval, although it may be possible to leave the fan to every third shop visit.

As with engines operated on short EFC times of 2.8-4.5EFH, shop-visit costs will be \$5.5-6.5 million where just the core and possibly a light workscope on the LPT is performed. Larger workscope that include the fan and LPC modules would cost \$6.5-7.5 million. On the basis that the engine can be managed so that the fan and LPC module is worked on every third shop visit, total costs will be \$20-22 million. The total interval for the three shop visits will be 40,000-48,000EFH. Reserves for these shop visit inputs will therefore be \$455-500 per EFH (see table, this page).

The second main element of maintenance reserves is for LLPs for all groups of LLPs. As described, this is a rate of \$445 per EFC. This is equal to \$74 per EFH when considered at the engine's EFC time of 6.0EFH.

Aircraft operating at this FH:FC ratio will not accumulate 30,000FC in their operational lifetime. If reserves for fan blades and cases are avoided, then reserves for LLPs will be reduced to \$350 per EFC. This is equal to \$58 per EFH when considered at the engine's EFC time of 6.0EFH.

The total for these two elements is \$518-558 per EFH (see table, this page).

## GE90 STANDARD SHOP VISIT REMOVAL INTERVALS & MAINTENANCE RESERVES

EFH:EFC Ratio	2.8-4.5	6.0	8.0
<b>1st removal interval</b>			
EFC	3,500	2,700-3,000	2,300-2,800
EFH	12,250-14,000	16,000-18,000	18,500-22,400
LLP limiter:	3,500EFC-HPT module	3,500EFC	3,500EFC
Shop visit input:	\$5.5-6.5 million	\$6.0-6.5 million	\$6.0-6.5 million
<b>2nd removal interval</b>			
EFC	4,500-5,500	2,200-2,700	2,000-2,300
EFH	16,000-22,000	13,000-16,000	16,000-19,000
LLP limiter	9,500EFC-HPT module	N/A	N/A
Shop visit input:	\$6.5-7.5 million	\$6.5-7.5 million	\$6.5-7.5 million
<b>3rd removal interval</b>			
EFC	4,500-5,500	2,200-2,700	2,000-2,300
EFH	16,000-22,000	13,000-16,000	16,000-19,000
LLP limiter	Various: 12,300-14,600EFC	9,500EFC	N/A
Shop visit input:	\$6.5-7.5 million	\$6.5-7.5 million	\$6.5-7.5 million
<b>Total interval</b>			
EFC	12,500-14,500	7,100-8,400	6,300-7,400
EFH	44,500-58,000	40,000-48,000	50,500-60,400
Total shop visit inputs:	18.5-21.5 million	\$19.0-22.0 million	\$20.0-22.0 million
Shop visit reserve:	\$325-385/EFH	\$460-500/EFH	\$385-420
LLP reserve:			
\$/EFC	\$445	\$445	\$445
\$/EFH	\$100-160	\$58	\$44
Total shop visit & LLP reserve-\$/EFH	%425-545	\$518-558	\$429-464

The main reason this is higher than engines operated at EFC lengths of 2.8-4.5EFH is because the first removal intervals were limited.

The cost of unscheduled removals and QEC should be considered in addition.

### Engines at 8.0EFH

Engines operating at longer average EFC times of 8.0EFH should be able to achieve slightly longer EFH intervals. First intervals were 2,300-2,800EFC, equal to 18,500-22,400EFH. The HPT interstage seal with a life of 3,500EFC would prevent a longer interval.

Second and subsequent planned intervals for the -85B, -90B and -94B have been 2,000-2,300EFC, which is equal to 16,000-19,000EFH. The total accumulated time would therefore be 66,000-80,000EFH by the fourth removal and shop visit, without an interruption due to unscheduled removals. This is equal to 8,200-9,600EFC. The fourth shop visit would therefore be forced by LLPs with limited lives.

Given that the second or third shop visits would require work on most

modules, a heavy visit and full disassembly would be required at either of these to prevent compromising the subsequent pattern of shop visit worksopes.

Most engines will be at mature intervals and maintenance status. They will follow the same or similar workscope pattern over three worksopes to engines operated at 6.0EFH per EFC. Engines operated at 8.0EFH per EFC will achieve slightly longer intervals, and will therefore have lower reserves for shop visits because the costs will be amortised over a total interval of 50,000-60,000EFH. Reserves will therefore be \$385-420 per EFH (see table, this page).

The second main element of rotating LLPs will be a reserve of \$350 per EFC, equal to \$44 per EFH (see table, this page). Total reserve for the two main elements is \$440-476 per EFH.

### GE90-110B/-115B

The GE90-110B and -115B power ultra-long-range aircraft. The majority of engines operate on long average FC times of 7-11FH, although a minority operate on shorter FC times on medium-haul

## GE90-110B/-115B SHOP VISIT REMOVAL INTERVALS &amp; MAINTENANCE RESERVES

Engine EFH:EFC ratio	-115B (777-300) 8.5	-110B (777-200LR) 10.0
<b>1st removal interval</b>		
EFC	2,900-3,100	2,800-3,000
EFH	25,000	28,000-30,000
LLP limiter:	3,500	N/A
Shop visit input:	\$6.0-6.5 million	\$6.5-7.0 million
<b>2nd removal interval</b>		
EFC	2,200	2,000
EFH	19,000	20,000
LLP limiter	N/A	N/A
Shop visit input:	\$7.0-7.5 million	\$7.5-8.0 million
<b>3rd removal interval</b>		
EFC	2,200	2,000
EFH	19,000	20,000
LLP limiter	N/A	6,780-HPT CASE
Shop visit input:	\$8.0 million	\$8.0 million
<b>Total interval</b>		
EFC	7,300-7,500	6,800-7,000
EFH	63,000	68,000-70,000
Total shop visit inputs:	\$21.0-22.0 million	\$22.0-23.0 million
Shop visit reserve:	\$333-350/EFH	\$325-330/EFH
LLP reserve:		
\$/EFC	\$432	\$432
\$/EFH	\$51	\$43
Total shop visit & LLP reserve-\$/EFH	\$483	\$475

operations.

The GE90-110B/-115B have identical components, but are rated at two different thrust ratings. Like the GE90 Base engine, the 128-inch-fan engine does not generally have engines removed due to EGT margin loss. Most removals are driven by hardware deterioration.

THY had its first engines delivered in 2010. "Initial installed EGT margin is 35-49 degrees centigrade, and so an average of about 43 degrees," says Erkan Evcan, CF6 and GE90 engine programme manager at Turkish Technic. "The rate of EGT margin loss in the first 1,000EFC on-wing is about 12 degrees. This settles to about five degrees per 1,000EFC thereafter."

This amount of initial EGT margin is therefore enough for the engine to remain on-wing for up to about 5,500EFC. "We actually expect the first planned removal intervals to be about 3,500EFC. We operate at about 9EFH per EFC, so this is equal to more than 30,000EFH," says Evcan. "This is for the later-built engines that do not have any of the technical problems that the earlier-built engines had."

Early production 128-inch-fan

engines have had a series of technical issues that have forced removals. "The three main issues that caused problems were the LPT stage 6 blades, the HPT blade shroud, and the leaf seal in the last stage of the HPC," says Uwe Zachau, senior manager performance & central engineering at MTU Maintenance. "All of these forced early removals, which meant that a lot of shop visits took place earlier than planned. This disrupted the shop-visit plan and pattern. We are expecting the first planned shop visits in 2013 for later-build engines that have had modifications to overcome these issues. These first planned visits will be core and performance restorations following long first-run removal intervals."

The problem of LPT stage 6 blades was that some separated and broke lose from the turbine disk on eight different occasions. These separations could lead to uncontained engine failures, so an airworthiness directive (AD) was issued. AD 2009-25-14 requires repetitive inspections if two particular LPT stage 6 blade part numbers are installed. These inspections are then terminated if new blades are installed.

The inspections are required prior to

3,000EFH or 400EFC being accumulated to detect wear of the shroud interlock. Repeat inspections are then required once every 1,000EFH or 125EFC, whichever comes first. Replacement of the two part numbers with new blades terminates the need for inspections.

A second major problem was the liberation of small parts from the seal teeth between stages 1 and 2 of the HPC. AD 2011-26-11 was issued and required eddy current inspection or other non-destructive testing (NDT) of the seal teeth.

There was also an issue with the HPT shroud, which required a change of HPT blades.

Zachau comments that in some cases these issues could be resolved through a minor quick turn repair, to fix the problem, and a full shop visit would not necessarily be required.

The problem of the pin coming loose that holds the leaf seal in place at the last stage of the HPC caused the seal to separate. This resulted in a leak of airflow into the HPC drum, which increased EGT and so reduced engine efficiency. General Electric recommends a borescope inspection at this point. "There have been four fixes for this leaf seal problem," says Zachau. "These have been detailed in service bulletins (SBs). SB numbers include 72-337/-338. Once the problem has been detected the affected parts need to be replaced or repaired. SB 72-424/-425 describes the repair process to introduce the new hardware."

SB 72-337 details performance trend monitoring and the inspection of the stage nine outlet guide vane (OGV) leaf seals. SB 72-988 details a rework to add a longer leaf seal pins to the stage 10 OGV assembly. Other SBs issued in relation to the leaf seal problem are SB 72-290, SB 72-354 AND SB 72-487.

"Premature removals of engines due to these technical issues have been very short: as little as 100EFC and up to 1,500EFC," continues Zachau. "There have so far been about 300 shop visits globally to deal with leaf seal problems. Some engines have had two shop visits to cope with the issue.

"If the problem happens after an interval of 1,500-1,800EFC then a performance restoration can be carried out, although normally this would not be expected until at least 2,000EFC had been accumulated," explains Zachau.

Evcan also highlights the issue of the rubbing of third and fourth stage HPC blades on the inner compressor wall. "This can fortunately be dealt with by using a quick turn repair, and does not require a full shop visit," says Evcan.

Once early-build engines had been cured of their initial reliability problems, mature removal intervals would settle down to 2,000-2,200EFC and 18,000-



20,000EFH. Like the GE90 standard, the 128-inch-fan engines would conform to a similar shop-visit pattern. This would involve all HP modules at every shop visit, and work on the fan and LPC every second or third shop visit. Work in the LPT could be at alternate shop visits, with the second shop visit having a light LPT workscope, and the fourth having a full workscope. A full LPT workscope may have to be included at the fourth shop visit.

Younger engines, that have not experienced the reliability problems that the older engines experienced, are generally expected to be able to achieve a full planned first removal interval of about 2,800-3,000EFC. At EFC times of 8.5EFH and 10.0EFH, this is equal to 25,000EFH and 30,000EFH.

The subsequent second and third intervals are expected to be 2,000-2,200EFC and so 19,000-20,000EFH.

The total accumulated time over three shop visits is thus expected to be 63,000-70,000EFH and 6,800-7,500EFC (see table, page 16).

The costs for the first shop visit will be \$6.0-6.5 million. Taking into consideration experience of the GE90 standard, costs for the second and third shop visits are expected to be \$7.0-7.5 million each. The third shop visit could rise closer to \$8.0 million. The total cost of these three shop inputs is therefore

\$21.0-23.0 million.

A reserve of \$333-350 per EFH for engines operated at 8.5EFH per EFC, and \$325-330 per EFH for engines operated at 10.0EFH per EFC should therefore be budgeted for (see table, page 16).

Consideration also has to be given to reserves for LLPs. There are three elements of rotating parts, fan blades, and cases.

A shipset of rotating LLPs has a list price of \$6.5 million. These have target lives of 15,000EFC.

Assuming compensation for those parts with restricted lives is possible, or that these parts have their certified lives extended, reserves would be close to the list price amortised over the target life of 15,000EFC. This would be equal to \$432 per EFC.

Each fan blade has a list price of \$116,000 and a life limit of 30,000EFC. Reserves for a full shipset would therefore be \$85 per EFC.

Although these parts have a fixed life, they are susceptible to foreign object damage or delamination, which can incur a high repair cost.

The third element is for engine cases. These have target lives of 40,000EFC, although the six parts have lives restricted to 6,780-25,380EFC. Again if compensation for parts with restricted lives is possible, then a reserve for these parts with a combined list price of \$5

*The GE90 Standard and Growth variants should have reserves for shop visit inputs and LLPs in the region of \$425-560 per EFH.*

million would be \$125 per EFC.

As with the 123-inch fan engines, the life limits of fans and cases are unlikely to be reached during the aircraft's operational life. Reserves for LLPs could therefore be limited to the rotating parts, with a reserve of \$432 per EFC. This is equal to reserves of \$43 per EFH for the -110B engine operated at 10.0EFH per EFC, and \$51 per EFH for the -115B engine operated at 8.5EFH per EFC (see table, page 16).

Total reserve for shop visits and rotating LLPs is thus \$475 per EFH for the -110B engine operating at 10.0EFH per EFC, and \$483 per EFH for the -115B engine operating at 8.5EFH (see table, page 16).

## Maintenance services

Many GE90 engines are maintained under General Electric's (GE) OnPoint services. These include shop visit maintenance, maintenance management, engine health monitoring (EHM) and range of other support products. Operators can sign contracts for OnPoint services for a particular period, and then decide whether to renew or seek alternative maintenance support providers.

The other providers of maintenance support for the GE90 include MTU Maintenance, Air France Industries KLM Engineering & Maintenance, Taexl in Xiamen, All Nippon Airways and Emirates. MTU maintenance is the only provider that is completely independent to GE.

Besides opting for time and material contracts with MTU Maintenance for the 128-inch fan engine, airlines can also subscribe to its EHM, shop visit planning and workscope definition, hi-tech parts repairs planning, and access to a pool spare engine.

"We offer time and material, never-to-exceed (NTE) cost maintenance, and power-by-the-hour (PBH) contracts," says Zachau. "The hi-tech parts repairs we will offer may become OEM-licensed, since many lessors prefer these to non-licensed."

MTU Maintenance also offers line maintenance aircraft-on-ground (AOG) support. [AC](#)

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